

### **SB AD32/06**

Revision: 1.0

TITLE:	Hardware Modification for AD32 Air Data Display (HIRF compliance according to JAA INT/POL/27&29/1)	
DOCUMENT NUMBER:	SB AD32/06	
EQUIPMENT:	AD32 Air Data Display	
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## **RECORD OF REVISIONS**

Rev.	Date	Reason for Revision	Prepared	Checked	Approved
1.0	04/09/19	Initial Release	J. Garrett	Mhort H. Richardt	A. Anehila



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### **SERVICE BULLETIN**

### Planning Information

### A. Effectivity

The modification procedure described in this Service Bulletin applies to the AD32 Air Data Display with HW 4.10.

### B. Concurrent Requirements

None required.

#### C. Reason

This document is issued to provide information about the hardware change for a higher HIRF protection (in accordance with JAA INT/POL/27&29/1).

### D. Description

The following hardware modification must be done to accomplish the higher HIRF protection, in accordance with JAA INT/POL/27&29/1:

Replacement of the main housing, the front glass, the centering ring and its O-ring seal.

Addition of copper contact spring(s), located between the main housing and the axle(s) of the operating knob(s).

Replacement of the MOD status and identification labels.

### E. Compliance Recommendation

This Service Bulletin is to introduce improvements for a better HIRF protection. Accomplishment is optional.

### F. Approval

The AD32 Air Data Display conforms to TSO-C106, TSO-C88a, and TSO-C10b.

This Service Bulletin contains no modification information that revises the approved configuration and therefore does not require any implementation of governmental or other regulatory agency approval.

### G. Manpower

This modification can be done by THOMMEN AIRCRAFT EQUIPMENT AG or its approved service centres.

The modification (excluding removal, reinstallation of the unit and acceptance test) requires approximately one man-hour.

### H. Weight and Balance

Not affected.

### I. Electrical Load Data

Not changed.

### J. Software Accomplishment Summary

Not applicable.



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### K. References

REFERENCE	ITEM	
WIN-0010-002	Work Instructions - AD32	
CMM-34-16-10	Component Maintenance Manual - AD3x.()	
IPL-34-16-10	Illustrated Parts List and List of Special Executions - AD3x.()	
051010AFB	Test Report (done by EMCC Dr. Rasek)	
SR AD32/01	Similarity Report for AD32 Air Data Display (HIRF Compliance)	

### L. Other Publications Affected

REFERENCE	ITEM	
AD-DDP-400	Declaration of Design and Performance	
AD-INSOP-400 Installation and Operation Manual - AD3x.()		

## M. Interchangeability of Parts

EXISTING PARTS		REPLACEMENT PARTS	
ITEM	PART NUMBER	ITEM	PART NUMBER
Housing	211623313	Housing	211623333
N/A	N/A	Contact spring(s)	211740914
Centering ring	211593414	Centering ring	211593424
Glass assembly	210716414	Glass assembly	211737914
O-ring	35526303	O-ring cord (230mm)	65223019

All other parts are unaffected by this Service Bulletin.



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### 2. Material Information

### A. Material - Price and Availability

The hardware parts/materials required for this modification are procured by THOMMEN AIRCRAFT EQUIPMENT AG or its approved service centres.

Installation of this Service Bulletin is subject to no special pricing.

For more information please contact:

THOMMEN AIRCRAFT EQUIPMENT AG Technical Support Department Hofackerstrasse 48 CH-4132 Muttenz Switzerland

Tel: +41 (0)61 965 22 22 Fax: +41 (0)61 965 22 88

NOTE:

All units subject for modification must be scheduled prior to shipping. Contact THOMMEN AIRCRAFT EQUIPMENT AG for further lead time and delivery schedule.

### B. Tooling - Price and Availability

No special tools are required.



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### 3. Accomplishment Instructions

The modification procedure can be accomplished by THOMMEN AIRCRAFT EQUIPMENT AG or its approved service centres.

### A. Preparation

- (1) Remove the AD32 from the aircraft.
- (2) Make sure that the workstation is clean and clear of unwanted parts and materials



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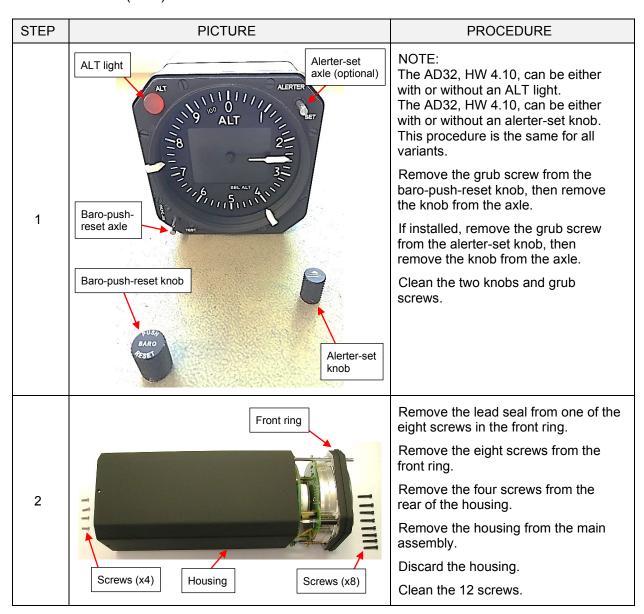
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#### B. Procedure



CAUTION:

THE PRINTED CIRCUIT BOARDS (PCB) INSIDE THE AD32 ARE ELECTROSTATIC DISCHARGE SENSITIVE (ESDS) PARTS. YOU MUST MAKE SURE THAT THE WORKSTATION IS ELECTROSTATIC DISCHARGE (ESD) COMPATIBLE BEFORE DOING THE FOLLOWING PROCEDURE.



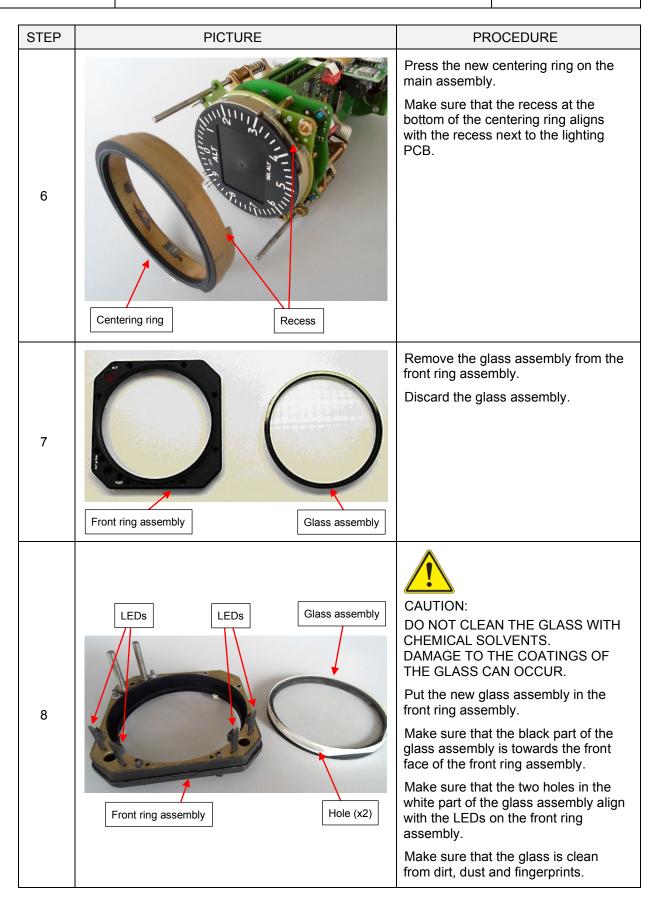


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STEP	PICTURE	PROCEDURE
3	Lighting PCB Springs (x2)	Carefully pull the front ring assembly to disconnect it from the lighting PCB.  Remove the front ring assembly and the two springs (for the scale and test buttons).  NOTE: The AD32, HW 4.10, can be either with or without a scale button. This procedure is the same for both variants.
4	Front ring assembly  Main assembly  Centering ring	Carefully pull the centering ring from the main assembly.  Discard the centering ring and its O-ring.
5	Centering ring  O-ring co	Press the new O-ring cord (230mm long) in the groove of the new centering ring.  Cut the O-ring cord to the correct length, if necessary.

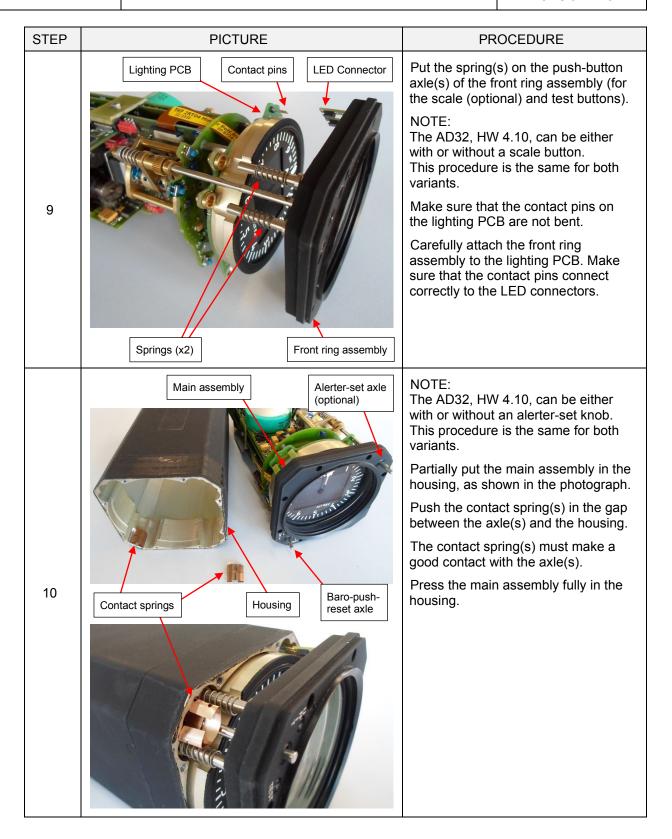


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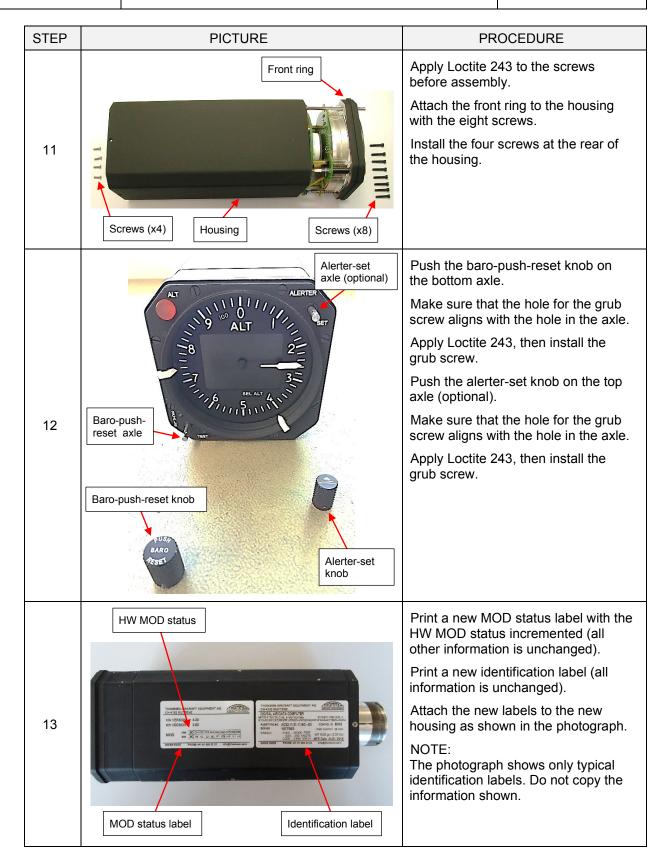


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## C. Close Up

Do an acceptance test in accordance with the applicable Acceptance Test Procedure (ATP) and the associated Acceptance Test Record (ATR).

All testing will be carried out by THOMMEN AIRCRAFT EQUIPMENT AG or its approved service centres.